

# -AIRCRAFT LOG

ASA-SA-2

Manufacturer Beech (ruft	N	lodel F33A	
Serial CE-1235	R	egistration Number_	N1235C
Date of Manufacture			
Engine(s) currently installed:			
Manufacturer	Model	Seri	al
Manufacturer	Model	Seri	al
Propeller(s) currently installed:			
Manufacturer	N	Model	
HUB Model	Serial	Seri	al
Blade Model	Serial	Serial	Serial
Blade Model	Serial	Serial	Serial

	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Reparenties must be endorsed with Name, Rating and Technician or Repair Facility. (See back pages for	Certificat
	Repa	ir Stat		DRA AIRPORT/SUGAR GROVE, IL 60554 BRO29L AIRFRAME LOG	
	MANUFACTURE	EO.	79704.0	R 649304R	
T	The Aircraft or App	ations of t	he Federal A	e was repaired and inspected in accordance Aviation Administration and is approved for repair are on file at this repair station under	D
re	eturn to service.	r entillenit c			
O	order No			10915 1EW For 15	NT.
O Da	order No	s	igned Le		NT.

N # 12350 REC. HR. METER 1025 S/N CE-1235 INSTALL OPTIONAL 100 AMP ALTERNATOR: REMOVED 60 AMP ALTERNATOR F/N 646843 S/N G089226.0 AND INSTALLED OPTIONAL DVERHAULED 100 AMP ALTERNATOR P/N 549304R S/N E079704. OR FROM T&W ELECTRICAL SERVICE, REF. W.O. 10915 PER THE BEECHCRAFT F33A WIRING DIAGRAM. GROUND RUN OPERATIONAL CHECK WAS SATISFACTORY. WEIGHT & BALANCE DATA WAS UPDATED. INSTALLED WIRING, CONNECTOR AND CIRCUIT PROTECTION PROVISIONS IN BAGGAGE COMPARTMENT FOR CUSTOMER EQUIPMENT. ALL WORK ACCOMPLISHED IN ACCORDANCE WITH AC43.13-18/24. SECTIONS 6, 8, 11 & 15. THE AIRCRAFT, ENGINE, APPLIANCE, OR COMPONENT IDENTIFIED WAS REPAIRED & INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE FEDERAL AVIATION AGENCY AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR ARE ON FILE AT THIS REPAIR STATION UNDER W.D. # 17871 DATE: JANUARY 26, 2010 For J.A. AIR CENTER, AURORA AIRPORT, SUGAR GROVE, IL 60554 F.A.A. APPROVED REPAIR STATION #NF2R029L

J A AIR CENTER AURDRA AIRPORT/SUGAR GROVE, IL 60554 Repair Station #NF2R029L AIRFRAME LOG N # 12350 S/N CE-1235 REC. HR. METER 1025 INSTALL CABIN DOOR STEWARD: INSTALLED MOUNTAIN VIEW AVIATION CABIN DOOR STEWARD P/N BC40C10 IN ACCORDANCE WITH STC SA1120SE. DRAWINGS LISTED ON APPROVED MODEL LIST SA1120SE DATED SEPTEMBER 9. 2002. DR LATER APPROVED REVISION AND INSTALLATION INSTRUCTIONS MVA-B40C10M&D REVISION A. DATED AUGUST 10. 2009. WEIGHT & BALANCE DATA UPDATED AND FAA FORM 337 COMPLETED. THE AIRCRAFT, ENGINE, APPLIANCE, OR COMPONENT IDENTIFIED WAS REPAIRED & INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE FEDERAL AVIATION AGENCY AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR ARE ON FILE AT THIS REPAIR STATION UNDER W.O. # 17872 DATE: JANUARY 26, 2010 SIGNED: Dall Mary J.A. AIR CENTER, AURDRA AIRPORT, SUGAR GROVE, IL 60554 F.A.A. APPROVED REPAIR STATION #NF2R029L

YEAR 20 DATE	TACH TIME	FLIGHT	TIME IN SERVICE	Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
3/1/10	1016.9		Removed See 337	GEM-610. Installed JP1 EGT-710. form dated 3/1/10, fold. Riche 262536133. TH
9/27/10	1098.2		Replace right	Main and mose tire/whell assemblies.  RICH 262536133 A4P

# Switlik Aviation Maintenance, Inc.

Page Field Airport (FMY) 207 Danley Drive Fort Myers, Florida 33907

239-936-3666 www.switlikaviation.com

C33-330-3000	AA AA AA 'S AA ITIIK'S	aviation.com			// <del></del>		
Date	Reg. number	Airframe S/N	Hour meter	Total		****	*****
12/03/2010	N1235C	CE-1235	1105.7	1105.7		****	*****

Lubed landing gear. Greased wheel bearings. Replaced left outboard brake pads. Replaced nose gear fork & axle assembly with p/n 35-825195-79. Replaced ring, scraper, and seals in nose strut and serviced with hydraulic fluid and nitrogen. Checked landing gear gearbox fluid. Checked landing gear normal and emergency operation. Checked landing gear dynamic brake function. Inspected shimmy dampner. Lubed flight controls. Cleaned fuel screen. Serviced tires, battery, and brakes. Inspected ELT per 91.207 (d), battery due 8/2011. I have inspected this airframe in accordance with an annual inspection and find it to be in airworthy condition.

John Rice

John A Richardson 262536133 IA

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
Secured as The aircraft found AIRV 2/17/2011 Carter Airc	ntray in the right and/or compoint VORTHY for ret ROGER E. SMI raft. Inc. FAA A	t hand dod nent identifi um to serv TH oproved Re	ed above was rece. Pertinent d	
Prop - Type Drained all o contaminati the engine a The aircraft found AIRW 2/17/2011 R	S/N Time of the hot oil out on was found in inspected for and/or components	of the engine side. Filled it oil leakagent identifier in to service the service of the service	ne sump. Rem the engine with e. No oil leaka d above was re the Pertinent de	oved the oil filter and cut open. Inspected the oil filter element and no metal Phillips X/C W100-50Wt oil and installed a new Ch48109-1 filter. Test range is evident at this time. Paired and inspected in accordance with current FAA regulations and was stails of this repair are on file at this agency under Work Order Number 7604.
Adjusted the 14 in/Hg. Re- of the open of Checked for The aircraft a found AIRW	throttle gear we moved the fuel uel tank nipple to fuel leakage and ind/or compone	arning switch quick drain to remove of d no leakage nt identified	from the right ther possible of the noted around above was re	BA: S/N: CE-1235: Tach: 1159.80. Total Time: Irrement that the aircraft operator established in the air to be the equivalent of fuel tank and removed small fleck of white paint from the seal. Flushed fuel out contaminates from the tank and installed the cleaned quick drain valve of this quick drain.  paired and inspected in accordance with current FAA regulations and was tails of this repair are on file at this agency under Work Order Number 7625.  JW4R585M.  pring FL.

DATE	TIME	SERVICE	TOCHINGIAN OF TOPAN ASSESSMENT OF THE PARTY
Installed Installation Report 6i checked the charg Aircraft F Garmin s The aircraft found Alf 3/18/201	a P2 P/N 6601 / th Instructions. It 500 Rev L Sec in accordance with ing system outp light Manual Sur- creens and instruction of the system output creens and instruction of the system output creens and the system output cr	Audio Advisory System in Report 6600. Rev L. Date ion Reference 4, Table F ith AC 43. 13-1B. Chapter jut. Modified the aircraft w oplement Audio Advisory alled screen protectors. As opent identified above was	F33A: S/N: CE-1235; Tach: 1163.60: Total Time: accordance with STC011114CH. (AML) No. SA01114CH. and P2 Hardware ad 8/10/2006. Tested the installed system in accordance with check list in P2 Titled AAS System Operational Check. The new total electrical load was 11. Section 3, paragraph 11-33, and 11-36(b) and found not to exceed 80% of eight and balance and the aircraft equipment list. Attached the FAA Approved System Per STC SA01114CH to the aircraft POH and Flight Manual. Cleaned the djusted the tightness of both of the aft elevator push rod bolts. Is repaired and inspected in accordance with current FAA regulations and was details of this repair are on file at this agency under Work Order Number 7630.  NUW4R585M.  Sebring FL.
assembly Engineerin INSTALLA System in L. Dated 8/	Whelen upper A Model 7090005 g Company. Inc TION AND SER accordance with 10/2006. Tested	nti-Collision rotating beac P/N 01-0770900-05. This INSTALLATION GUIDE VICE MANUAL Dated M STC0111114CH, (AML) N	33A; S/N: CE-1235; Tach: 1163.60; Total Time: on light P/N 10-0790081 and installed Whelen LED flashing Anti-Collision light is light assembly was installed in accordance with STC SA615EA and Whelen FORM NUMBER 13828D AND WHELEN ANTI-COLLISION LIGHT SYSTEMS May 2005, Document No. 05131, REV 1. Installed a P2 P/N 6601 Audio Advisory No. SA01114CH, and P2 Hardware Installation Instructions, Report 6600, Rev Accordance with check list in P2 Report 6600, Rev: L Section: Reference 4.

Advisory System Per STC SA01114CH to the aircraft POH and Flight Manual. Cleaned all Garmin screens and installed screen protectors. Replaced the plastic door pull cup in the right arm rest.

The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found AIRWORTHY for return to service Pertinent details of this repair are on file at this agency under Work Order Number 7630 UW4R585M.

Table F Titled AAS System Operational Check. The new total electrical load was checked in accordance with AC 43.13-1B. Chapter 11. Section 3. paragraph 11-33, and 11-36(b) and found not to exceed 80% of the charging system output. Modified the aircraft weight and balance and the aircraft equipment list. Attached the FAA Approved Aircraft Flight Manual Supplement Audio

Carter Aircraft. Inc. FAA Approved Repair Station. Sebring FL.

DATE	TIME	SERVICE	Technician or Repair Facility. (See back pages for other specific entries.)
Removed a aircraft arm The aircraft found AIRV 4/21/2011 5	rests. Adjusted the and/or component (ORTHY for return ROGER E. SMITH	e aircraft and installed P2 landing gear warn	3A, S/N: CE-1235: Tach: 1184.40: Total Time: hard placards, supplied by the aircraft owner, over the ash tray openings in the ing system speed up 5 knots per the aircraft owners request. epaired and inspected in accordance with current FAA regulations and was etails of this repair are on file at this agency under Work Order Number 7657.  UW4R585M. ebring FL.
	24 24 AH. A	PIPUED AK 12, SERUAL AUTED ) OPS SUHA . H	Date 04/24/11 Hours 1191.6  BATTERY W/ MEW G-  # 602538984. (SEX 8136  TEST W/ SATTSFACTORY  A Humberto Talavera 2777707 A&P  tenance Inc Fort Myers, FL 239-936-3666
Detail	ORLANDE	090° 00 VOC	Seach F3)A · VOR check - Bearing Tol #1 + # 2 0° difference - Thut I Esteld P.P.

Entries must be endorsed with Name, Rating and Certificate Number of TACH FLIGHT TIME IN Technician or Repair Facility. (See back pages for other specific entries.) TIME SERVICE Switlik Aviation Maintenance, Inc. Page Field Airport (FMY) 207 Danley Drive Fort Myers, Florida 33907 239-936-3666 www.switlikaviation.com Reg. No. S./N. Hobbs Date T.T. S.M.OH. Page N1235C CE1235 7/26/2011 Tach 1276.2 Installed new brake pads RH brake caliper. Bled R&L brake calipers. Had normal brake operation. Installed dual control wheel assy and rigged for proper operation. **Edward Switlik** IA2197076 Tomlinson Avionics of Florida, Inc. 521 DANLEY DRIVE FORT MYERS, FL 33907 V6TR097X LOG ID# 55 19-August-2011 N1235C S/N CE-1235 BEECH F33A Pg 1/1 REMOVED KING AUDIO SELECTOR KMA24 AND SOFT COMM INTERCOM AND INSTALLED PMA8000BT (AUDIO PANEL). SYSTEM CHECK PER MAN. SPECS.

RECORDING TODAY'S **Description of Inspections, Tests, Repairs and Alterations** TOTAL YEAR TACH FLIGHT TIME IN Entries must be endorsed with Name, Rating and Certificate Number of TIME SERVICE Technician or Repair Facility. (See back pages for other specific entries.) DATE Date: 9/03/2011; Aircraft: N1235C; Type: BEECH F33A; S/N: CE-1235; Tach: 1301.60; Total Time: Removed the aileron frim assembly and removed springs that had slipped out of the slots on the inside of the outer drum. Installed the springs correctly and installed the trim mechanism in the aircraft. Tested the aileron trim and found it to function properly. The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found AIRWORTHY for return to service Pertinent details of this repair are on file at this agency under Work Order Number 7769. 9/3/2011 ROGER E. SMITH KYZ/D UW4R585M. Carter Aircraft, Inc. FAA Approved Repair Station, Sebring FL. Date: 10/26/2011; Aircraft: N1235C; Type: BEECH F33A; S/N: CE-1235; Tach: 1333.40; Total Time: Jacked the aircraft up and removed both of the main wheels. Disassembled the wheels and inspected both wheels for hidden damage. Found no damage on either wheel. Replaced both of the tires with Flight Champion P/N 706C86-3 and assembled the wheels in accordance with the appropriate maintenance manual for this equipment. Both wheel assemblies were then balanced using T.J. Karg tire balancing equipment P/N 755-1 and their instructions. Installed the balanced wheel assembles back on the aircraft and cycled the landing gear to make sure that the new tires do not hang up in the wheel well. Both tires cleared the wheel well and the landing gear functioned normally. Removed the left brake caliper and disassembled it. Cleaned the caliper and the pistons in solvent. Inspected the caliper bores and the piston and found both to be serviceable. Installed new O-ring seals on the piston, lubricated the caliper and the seals and reinstalled the pistons in the bores. Installed the caliper on the aircraft and bled the brakes using MIL-H-5606 brake fluid. The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found ARWORTHY for return to service Pertinent details of this repair are on file at this agency under Work Order Number 7805. 10/26/2011 ROGER E. SMITH 4M. L UW4R585M.

Tomlinson Avionics of Florida, Inc. 521 DANLEY DRIVE FORT MYERS, FL 33907

LOG ID# 60 04-October-2011

N1235C S/N CE-1235 BEECH F33A

Carter Aircraft. Inc. FAA Approved Repair Station, Sebring FL.

Pg 1/1

REMOVED EXISTING CARPETING ON FLOOR AND SIDE PANELS. REPLACED WITH FAA APPROVED CARPETING (AC7314). REMOVED VINYL SIDE PIECES AND RECOVERED WITH FAA APPROVED VINYL. CARPETING AND VINYL REINSTALLED IN A/C TO ORIGINAL CONDITION. SEE BURN CERTS ATTACHED FOR FURTHER INFO.

Day 6 - 10-4-11

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
				Date 11/7/11 Hours 352.4  daileron trimasheel
	7	prings	ī	
				John A Richardson 262536133 A&P  enance Inc Fort Myers, FL 239-936-3666
415/11/2	NIESS C;	Buch F	33A	VOR check per FAR (891.171) - LAK-VOR: VOR 2 - 130° VOR 2 - 330°; O deviation The Hours S. Eskin P.P.
				ACR Electronics, Inc.  5757 Ravenswood Rd Ft. Lauderdale, FL 33312  954-981-3333  LOG BOOK ENTRY 2/17   2012  E.L.T. BATTERY  REPLACEMENT DATE 10/2013

A	oving National Aviation atherity/Country/ FAA/United States	AUTHORIZE FAA Form 81	D RELEASE C	ERTIFIC PROVAL TAG	ATE	5. Work Order/Contract/Invoice Number:
4. Organ	nization Name and Address: C&	M Marine Aviation Services, Inc. 00 King Arthur Drive				SO7870 (N1235C)
	Dal	llas, Texas 75247			- match	12. Status/Work:
6. Henr	7. Description:	nir Station #: V05R749Y 8. Part Number:	9. Eligibility: *	10. Quantili	11. Serial/Batch Number:	
1.	Restraint Restraint	501979-403 500779-9	n/a n/a	2 2	n/a n/a	Repaired Repaired
ebbing	g Flammability Requirem		10. 🖾 14	CFR 43 9 Deturn 10		Other regulation specified in Block 13
Critica	g Flammability Requirem	wannifectured in confermity income	19. 🗷 14	CFR 43.9 Return to ces that unless other scribed in Block 13 ations, part 43 and i	in a le Mor	13, the work identified in Block 12 secordance with Title 14, Code of Fede to teems are approved for return to
Central Centra	g Flammability Requirem	nents Meet 25.853(A)  Hunnfectured in conformity tree  Lorentees to the participation of the Mork 1.1	19. 🗷 14  Certifi and de Regule service	CFR 43.9 Return to ces that unless other scribed in Block 13 ations, part 43 and i	in a le Mor	21. Approval/Certificate No.  V05R749Y
Crinica Crinica	Planmability Requirem	nents Meet 25.853(A)  Hunnfectured in conformity tree  Lorentees to the participation of the Mork 1.1	19. 14 Certifi and de Regule service Authorization No.2 20. Author	CFR 43.9 Return to les that union other scribed in Block 13 ations, part 43 and i	wise specified in Bloc was accomplished in a respect to that work	accordance with Title 14, Code of Federal, the items are approved for return to  21. Approval/Certificate No.
Central Control	g Flammability Requirem	unnifectured in conferredly inc.  Contract in conferredly inc.  In Block 11.	19. 14 Certifi and de Regule service Authorization No.2 20. Author	CFR 43.9 Return to les that unless other scribed in Block 13 ations, part 43 and it. ized Senature: Typed or Printed): Brent Rol	wise specified in Bloc was accomplished in a respect to that work	21. Approval/Certificate No.  VOSR749Y  23. Date (m/d/y):

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DATE	HIME		SERVICE	Technician or Repair Facility. (See back pages for other specific entries.)
Completer	annual inspec	on check I	st this date on	33A; S/N: CE-1235; Tach: 1405.40; Total Time: 1405.40 this aircraftRemoved ELT 110-4, P/N 453-0150, S/N 68338, and inspected the
battery. Fo	und that the ba	ttery was n	bt leaking at thi	s time. Inspected the radio inside and found that the radio looks OK with no whattery that is good till October 2013. Tested the ELT system 5 minutes after
	the requirement	ate of contin	on Q1 207/d) an	LT manufacturer's instructions and applicable FAA guidance. Found that this d FAA action notice 8310.1. Removed the inlet filters for both of the pneumatic and the filter guards before installing new filters, P/N AAD9-14-5. Cleaned the
corroded a	reas on the bat	tery box co	ver. Primed and	AA2.14-7 Re-positioned the filter to prevent chaffing on the pneumatic valve
adjacent to	the filter. Sec	the should	te air cable ho	rtial reel assemblies from all 4 seats. Shipped the shoulder harness assemblies
out to C&N	Marine Aviatio	n Services	Inc. Repair St	s in the aircraft. Secured the aft interior light switch back in place in the plastic
	winer Cumlame	IC Sunvie	or kit R116000	flow molding. Removed the original sun visor system from the aircraft. Installed 5-0 in accordance with STC SA3598NM and Rosen Sunvisor Systems LLC egligible change in weight and balance. Modified the aircraft equipment list and
D-4-227	form with the E	AA coverin	nathic alteration	Tightened loose play in the nose gear steering mechanism on top of the nose and installed new brake lining. After installation of the lining the brake lining was
then condi	ioned in accord	ance with	he manufactur	on both fuel caps. Cleaned both main tank fuel quick drains and drain valves
seal prope	ly at this time.	rightened t	wo loose rivnul	s for the stinger installation. Removed the the right wing up strobe build assembly. CAV A-D 2007-08-04 by inspection of the propeller blades in
accordance	with McCaule	y ASB 248	2.B through 2F	Dressed blades per instructions. This inspection due again each armidal of
switch in a	cordance with	the operati	onal inspection	Called out in A-D 76-07-12 R1. Found the switch is all worthly at this time.
This aircra	t has been inspectalls of this ins	pected in a	cordance with on file at this	agency under work order number 7870.
2/17/2012 FAA Appro	Roger E. Smith ved Repair Sta	sion cane	Aircraft Inc. Se	bring Fl.

iu and Certificate Numb

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)

### ALTIMETER SCALE CORRECTIONS

Altimeter Mo	del 5934	HD-1	SN L	HE76+
Altimeter	Observed		Altimeter	Observed
Reads	Reading		Reads	Reading
-1000	0		12000	+30
0	0		14000	+45
500	0		16000	+30
1000	0		18000	+80
1500	0		20000	+100
2000	+5	1 1	22000	/
3000	+3		25000	
4000	+10		30000	
6000	+15		35000	X
8000	+35		40000	
10000	+30		45000	
	1.00		50000	

Tomlinson Avionics of Florida, Inc. 521 Danley Drive Fort Myers, FL 33907 Repair Station No. V6TR097X

## ER SCALE CORRECTIONS

,DC ++	A SN 2	2627000
rved	Altimeter	Observed
ing	Reads	Reading
)	12000	-10
)	14000	-10
7	16000	-10
5	18000	-10
5	20000	-10
5	22000	
5	25000	
	30000	
5	35000	
2	40000	
2	45000	
	50000	

YEAR

DATE

20

RECO

TIME

son Avionics of Florida, Inc. 521 Danley Drive ort Myers, FL 33907 in Station No. V6TR097X

Installed new vertical card compass P/N PAI-700 S/N 76722. Aligned compass on compass rose and made correction card for aircraft.
Signature Don
TOMLINSON AVIONICS OF FLORIDA, INC. PH.239-936-6969, 521 DANLEY, PAGE FIELD, FT. MYERS, FL. FAA APPROVED REPAIR STATION NO. V6TR097X
I CERTIFY THAT THE ALTIMETER, ENCODING SYSTEM AND STATIC SYSTEM TEST REQUIRED BY FAR 91.411 HAVE BEEN PERFORMED: THE ALTIMETER: PILOTS S/N. COPILOTS S/N. WAS/WERE TESTED TO K FT I CERTIFY THAT THE TRANSPONDER TEST REQUIRED BY FAR 91.413 HAS BEEN PERFORMED AND FOUND TO COMPLY XPDR # 1 M/N. S/N. SIGNATURE  DATE TOMLINSON AVIONICS OF FLORIDA, INC. 521 DANLEY DR., FT. MYERS, FL. 33907 FAA CRS # V6TR097X

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Removed the Whelen P/N installed using the original removed the gauge circulative aircraft found AIRW 3727/2012 Fe	e incandescent of the original loub. The weight eleft inboard for the original for the orig	landing lig 0, per FAA ight wires, it of the new pel sender a auges both ent identifier im to service H	ht bulb from the PMA PQ0610N switch and circle v bulb assemble and cleaned wif function proper d above was re- ces Pertinent de	3A; S/N: CE-1235; Tach: 1447.70; Total Time: a nose of the aircraft and installed a new landing light LED bulb assembly E, Whelen Supplement No. 37 dated December 2nd 2010. This bulb was ait protection. The new bulb draws only .6 amps which is substantially less than y matches the old bulb weight so there is no change in weight and balance. In precision clean. Removed the glare shield and tightened plugs in the fuel repaired and inspected in accordance with current FAA regulations and was tails of this repair are on file at this agency under Work Order Number 7917.  UW4R585M.  Dring FL.
Date: 2/17/20 Secured ashter The aircraft are found AfRWD 2/17/2011 RD Carter Aircraft	011; Aircraft: N ray in the right nd/or compone ORTHY for retur OGER E. SMIT I, Inc. FAA App	1235C; Typ hand door a nt identified in to service the control of the control in to service the control of the control of the control in to service	e: BEECH F33 ormrest using for above was re- pertinent detail	A; S/N: CE-1235; Tach: 1131.60; Total Time; pam tape. Adjusted the landing gear throttle horn switch position. paired and inspected in accordance with current FAA regulations and was alls of this repair are on file at this agency under Work Order Number 7604. W4R585M.
Date: 7/190 Inspected all Communique the instrument range of motion The aircraft are bound AIRWO 7/19/2012 RO	012; Aircraft: N 4 aileron hinge 1 324. That pain at panel per Har on by an assist	1235C; Types to verify part work will wher safety and to the interest to service the service of	pe: BEECH F3 proper screw er be performed a Communique aspector and no above was rep	A; S/N: CE-1235; Tach: 1499.90; Total Time: Ingagement. The aileron hinges were not painted per Hawker Safety It a later date. Complied with inspection of the elevator control cables behind 322 by visual inspection. The elevator control was moved through its full ordamage or corrosion on the specified cables was noted. Deaired and inspected in accordance with current FAA regulations and was ails of this repair are on file at this agency under Work Order Number 8020. W4R585M.

YEAR 20 DATE

DC min a R b b th T in p g is b o lch 2 5 a w a

20 1	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name. Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
light. The	d the left wing e light function solvent. Inspending abricated the	tip lens an is properly, ected the c	d replaced the Removed the raliper bore and	H F33A; S/N: CE-1235; Tach: 1531.70; Total Time: ed nav light lens. Installed the wing tip lens and checked the operation of the nav ight brake caliper from the wheel and removed the piston. Cleaned the caliper and found that it is still in serviceable condition. Installed a new O-ring seals on the tons in the caliper. Installed the brake on the aircraft and bled the brakes with
The aircrafound Alf 9/11/2012	aft and/or cor RWORTHY fo 2 ROGER E.	SMITH	service-Pekine	ras repaired and inspected in accordance with current FAA regulations and was not details of this repair are on file at this agency under Work Order Number 8058.  **UW4R585M.**
Date: 12/15 Completed manufactur inspection is air filters. P. Removed the both ends. I bandit claim this time. Re The pump to installed a re plastic shout gear position is worn out. brake lining compressed loose wire of hinge bracke 2333 and for 500 hours the air induction when it has	Annual inspenser's instructions due within I/N AAD9-14- the fuel line from installed the surned freely, the pump coulder harness in indicator surned freely, was conditiond Nitrogen gas wer the pilots ets from their und no crack me in service filter element accumulated.	off: N12350 oction check ins and FA in its and FA in its and in its and its an	Type: BEECH Ist this date of A action notice of the last inspense standard purvall to the medings on the endings on the purpose stalled the purpose stalled the purpose splug. Removed the left wing pedal. Complete left wing pedal. Compl	F33A; S/N: CE-1235; Tach: 1557.90; Total Time: In this aircraft. Inspected the ELT system in this aircraft according to the ELT 8310.1. Found that this ELT meets the requirements of section 91.207(d). This ection. The ELT batteries are dated Good Till October 2013. Installed two new inlet imp inlet housing and the second filter on the standby pump inlet housing. Inanical fuel pump inlet. Removed the original line and removed the fittings from is of the new hose. Installed fire sleeve on the hose and secured in place with led on the aircraft and the fuel system was checked for leaks. Found no leaks at in the electric drive motor and inspected the pump for any signs of internal failure, le freely and there is no sign of broken carbon coming out of the pump ports. In point the standby motor. Tested the pump and it functions properly. Replaced the less loops at the tot of the egress window frames. Replaced 5 pins in the landing loved the brake assembly from the right wheel and found that all of the brake lining less called out in the maintenance manual for this aircraft. After installation, the le manufacturers instructions. Set the landing gear oleo strut heights using lip strobe tube assembly. Tested the strobe light and it functions properly. Secured led with A-D 93-24-03 by removing the rudder from the aircraft and removing the rudder spar with dye penetrant in accordance with Beech Service Bulletin No. Is time. This A-D is recurring and must be repeated again at the accumulation of stallation of a new paper air filter. This A-D requires the replacement of all paper 500 hours time in service. This filter will require replacement at Tach Time 2057.9 Is or on condition as determined by inspection. Inspected the ignition switch in location A-D 76-07-12 R1. Found the switch is airworthy at this time. This inspection line in service.

Smith Sebring Fl. craft: N 23 g taxi light is a FAA Fal Informati C 43.13-16 out. Negligi JPI P/N 1.	Repair S 35C: Type PMA Apprison, Supp B, Chapte ible chan 1280. Test identified to service	Station.  e: BEECH F33 and installed roved replacer olement Number 11, Section age in weight at ran the engine above was replaced above was replaced by the control of the	A; S/N: CE-1235; Tach: 1588.40; Total Time: a replacement Whelen LED light assembly Model PAR36, P/N nent for the existing light and this aircraft is listed on the FAA Parts ar: 36, Dated 12/20/2010. The new total aircraft electrical load was checked 3, Paragraph 11-33, and 11-36(b) and found not to exceed 80% of the hd balance. Removed damaged EGT probe from cylinder number 5 and le and found that the exhaust gas temperature gage functions normally at this leaired and inspected in accordance with current FAA regulations and was alls of this repair are on file at this agency under Work Order Number 8146. 4R585M.
g taxi light is a FAA Fal Information C 43.13-16 out. Negligion JPI P/N 1. component infor return to MITH	P/N 4596 PMA Applion, Supple B, Chapte ible chan 280. Test identified o service	and installed roved replacer olement Number 11, Section age in weight at ran the engine above was replaced by the control of t	hent for the existing light and this aircraft is listed on the FAA Parts er: 36, Dated 12/20/2010. The new total aircraft electrical load was checked 3, Paragraph 11-33, and 11-36(b) and found not to exceed 80% of the hid balance. Removed damaged EGT probe from cylinder number 5 and e and found that the exhaust gas temperature gage functions normally at this paired and inspected in accordance with current FAA regulations and was alls of this repair are on file at this agency under Work Order Number 8146.
or return to	o service	Pertinent deta	ails of this repair are on file at this agency under Work Order Number 8146.
	ACCOUNTY OF	ing FL.	
35 C E	86-33		VOR Check - LAL (1160) NOR 1-355° VOR 2-355 O" deviation Tenully HARILDS Eaking Plot
load brack	ket arm.	be: BEECH F3 Shecked the ti	BA; S/N: CE-1235; Tach: 1578.20; Total Time: me since the last oil change and found that no oil change is due at this time. next.Installed "Garmin ADS-B Equipped" placard on the exterior of the
for return	do servis	e. Pertinent de	epaired and inspected in accordance with current FAA regulations and was tails of this repair are on file at this agency under Work Order Number 8153.  W4R585M.
	tating who	lpad blacket arm of stating when the of component identifier for return to service	tpad bracket arm. Checked the ti stating when the oil change is due component identified above was re

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YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
7	Tail A	11235	Date	6/10/13 Hours 1623.3
	In	tallock	del	control yoke.
	+	GIVE CS		- PRIMITY DIES
	- 61	110	in no	nn A Richardson 262536133 A&P
	1			ce Inc Fort Myers, FL 239-936-3666
10/13	NIZASC	31E-32		VOR CHECK X14 (LaBelle) 110.4 VOR 1-2530
				VOR CHECK X14 (LaBelle) 110.4 VOR 1-253° VOR 2 - 253° 0° Deviation
				HAROLD SESKIN PILOT
The aircr	aff and/or compo	nent identifi	ed above was	33A; S/N: CE-1235; Tach: 1610.60; Total Time: 32-103. The alternator functions properly at this time. epaired and inspected in accordance with current FAA regulations and was etails of this repair are on file at this agency under Work Order Number 8227. UW4R585M.
Carter Air	caft, Inc. Repair	Station, S	bring FL.	UW4R585M.
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YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
Date: 7/08/2 Removed broad-15. Approved Beblower assert The aircraft of found AIRWO 7/8/2013 RO Carter Aircraft Date: 8/0 Jacked the damage. Roasembled wheel asserbalanced with Beech nose Damper P/M new damper SE-1076-1 SRevision (Trange in we Checked the properly. Verlanding gear compressed itime Sarvice and the Sarvice street of the Sarvice street and the Sarvice	on the wheels using the wheel assemble shimmy damp is seen to escape.	entilation bencludes neighbors proper it identifies in a service station. Service N1235C; removed to be on either ing new when balance is back on the proper installed rebruary 0 ce. Cycled afety system with the station indical control of the station indicates the station indicates and the st	be: BEECH F33 lower assembly w blower assembly w blower assembly d above was re expertinent def porting FL.  Type: BEECH F loth of the main wheel. Replace bel bolt nuts in d using T.J. Ka he aircraft. Rep 3251445-2 from 7, as listed in the loce with Lord B On Beech 33, 1, 1, 2008 and the the landing ge m, position indi not hang up in lator. Removed hose wheel up	A; S/N: CE-1235; Tach: 1621.30; Total Time:  /, P/N 58-380108-1, and installed a factory new blower assembly kit, P/N hbly, P/N 58-380108-5 S/N 453, was installed in accordance with FAA is a minor alteration with negligible change in weight and balance. Tested the paired and inspected in accordance with current FAA regulations and was ails of this repair are on file at this agency under Work Order Number 8244.  W4R585M.  33A; S/N: CE-1235; Tach: 1641.70; Total Time: wheels. Disassembled the wheels and inspected both wheels for hidden accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance manual for this equipment. Both accordance with the appropriate maintenance direct replacement Lord Shimmy alletin No. SM-6523. Titled "Installation and Removal Instructions For Lord Bonanza 33, 35, 36, Maintenance Manual P/N 36-590001-9. Negligible cating system and the landing gear warning system and all functioned the wheel well. Replaced missing Velcro tape that secures the cover on the
nounted to the	e front bottom	er extensio comer. Ins	n. Found that to	lled the valve core. Filled the strut with dry Nitrogen in accordance with the his aircraft. Lowered the nose to the ground and moved the aircraft around to be extension is OK at this time. Straightened the left front cowling guide cowling catch mechanism and fond it to be functioning properly.  Tails of this repair are on file at this agency under Work Order Number 8270.

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DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)				
	Upgraded GTX 330 software to 7.02 per STC SA01714WI. For more information see 337 this date. Function check ok.  Signature  Date 7-16-13  TOMLINSON AVIONICS OF FLORIDA, INC. PH.239-936-6969, 521 DANLEY, PAGE FIELD, FT. MYERS, FL. FAA APPROVED REPAIR STATION NO. V6TR097X							
FAR 91 Service that the Replace	1 207. A function ad all tires in acc arm rest supported the missing s	praft: N1235 and model E nal ELT must cordance with ort had corn screws and ponent iden	C; Type: BEE ELT 110-4 wi st be installed th the Bonan e apart. Loca secured loos	CH F33A; S/N: CE-1235; Tach: 1649.90; Total Time: 1649.90 as removed for maintenance this date. Panel placarded "ELT NOT INSTALLED" per in this aircraft within 90 days for continued airworthiness of this aircraft. Plots Operating Handbook for this aircraft. Removed the pilots seat and discovered ted the missing hardware under the seat and reassembled the arm rest assembly. The screws in the baggage door wind lace.  Was repaired and inspected in accordance with current FAA regulations and was ent details of this repair are on file at this agency under Work Order Number 8323.				

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
		PERFORM COPILOTS WAS/WER I CERTIFY	E TESTED TO	REQUIRED BY FAR 91.411 HAVE BEEN  IMETER: PILOTS S/N 4F 16 16 16 16 16 16 16 16 16 16 16 16 16
AIL 1	ALTIMETER SO	CALE COF	RECTIONS	ADC ALTIMETER SCALE CORRECTIONS

ALTIMETER	SCALE	CORRECTIONS
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Altimeter Mo	odel 5934	AD-1	SN 44	=767
Altimeter	Observed		Altimeter	Observed
Reads	Reading		Reads	Reading
-1000	+5		12000	+50
0	+5		14000	140
500	+5		16000	+50
1000	+5		18000	+50
1500	0		20000	+40
2000	0		22000	
3000	+15		25000	
4000	+15		30000	
6000	+30		35000	X
8000	+35		40000	
10000	140		45000	/
-1		1 - 1	50000	

Tomlinson Avionics of Florida, Inc. 521 Danley Drive Fort Myers, FL 33907 Repair Station No. V6TR097X

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ft)	CDOT	110	0x 20	607562
Altimeter Mo	del 6DC	44		Observed
Altimeter	Observed		Altimeter	Series and the series of the s
Reads	Reading		Reads	Reading
-1000	+5		12000	-10
0	1<		14000	-10
500	+<		16000	-10
1000	6		18000	-10
1500	0		20000	-10
2000	0		22000	
3000	3		25000	
4000	8		30000	
6000	0		35000	
8000	0	1	40000	
10000	-10		45000	
10000	10	- L	50000	

Tomlinson Avionics of Florida, Inc.

521 Danley Drive Fort Myers, FL 33907 Repair Station No. V6TR097X

RECORD TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Entries must be endorsed with Name, Rating and Certifica Technician or Repair Facility. (See back pages for other sp								
	STATIC SY PERFORMI COPILOTS WAS/WERE I CERTIFY 91 413 HAS XPDR # 1 N XPDR # 2 N SIGNATURI	I CERTIFY THAT THE ALTIMETER, ENCODING SYSTEM AND STATIC SYSTEM TEST REQUIRED BY FAR 91.411 HAVE BEEN PERFORMED: THE ALTIMETER: PILOTS S/N PERFORMED: THE ALTIMETER: PILOTS S/N PERFORMED: THE ALTIMETER: PILOTS S/N PERFORMED AND S/N PERFORMED BY FAR 91.413 HAS BEEN PERFORMED AND FOUND TO COMPLY.  XPDR # 1 M/N FIRST S/N									

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
Inspected I that is good manufactur action notice regulations	Artex 110-4 ELT he radio inside I till February 2 er's instruction to 8310.1. The	serial nur and found 016. Tester and applic aircraft and Aisworthy	that the radio lo that the radio lo the ELT syste cable FAA guid / or component for Return to Se	33A; S/N: CE-1235; Tach: 1668.30; Total Time: 1668.30 I inspected the battery. Found that the battery was not leaking at this time. oks OK with no internal corrosion evident at this time. Installed a new battery m 5 minutes after the hour in accordance with applicable aircraft and ELT ance. Found that this ELT meets the requirements of section 91.207(d) and FAA i identified above was repaired and inspected in accordance with current FAA price.  A&P 3442201
Completed the ELT ma This inspecton pneumatic each year aircraft on and installe accordance cowl flap put the switch a The work of This aircraft Pertinent de 4/24/2015. Roger E. St.	Annual inspect inufacturer's instance in the second the valve corwith the decal ish rod. Inspects airworthy at the this annual whas been inspectable of this instance.	ion check I structions a hin 12 month from the ele sockets for the value assembling affixed to the ignitial time. The sperform ected in acceptation are	st this date on and FAA action his of the last in agine baffle and or the fuel selective cores from a ses. Pressurized the individual ution switch in a is inspection is ed in accordance with on file at this a UW4R5	his aircraft. Inspected the Artex ME406 ELT system in this aircraft according to notice 8310.1. Found that this ELT meets the requirements of section 91.207(d). spection. The ELT batteries are dated Good Till July 2020. Removed the installed a new filter, P/N AAD9-14-5. This filter requires replacement once tor lights completing the ground circuit. Lights function properly. Placed the II of the oleo stems. Serviced all of the oleos with X/C HY-5606A hydraulic fluid the struts with compressed Nitrogen and then adjusted the strut heights in pper strut tubes. Secured the fuel line with Ty-raps to stop interference with left coordance with the operational inspection called out in A-D 76-07-12 R1. Found due again upon the accumulation of 100 hours time in service. ce with the Beech 33, 35, 36 Maintenance Manual P/N 36-590001-9.  an ANNUAL Inspection and was determined to be in AIRWORTHY condition. gency under work order number 8676. This ANNUAL was completed.

Completed Annual inspection check list this date on this aircraft. Removed both of the brakes and rotors were conditioned in the practical process of the brake rotors. The brake rotors were conditioned in the brake rotors is not the brake and rotors were conditioned in the brake rotors. The brake rotors is not replaced an prake lining Hambfacture of the EDT make Artex and model EDT 1 to 4 Wagte Number of accordance with the districtions subfilled by the brake lining Hambfacture of the EDT make Artex and model EDT 1 to 4 Wagte Number of accordance with the districtions subfilled by the brake lining Hambfacture of the EDT make Artex and model EDT must be installed in order to be accordance with the manual part of the process of this aircraft. Replaced by the process is the first aircraft. Replaced by the process is the process of the process o this aircraft within 90 days for continued a pworthiness of this aircraft. Replaced both pneumatic pump inlet filters, P/N AAD9-14-5 and the instrument in-line filter, P/N AA2J4-7. Tightened loose hardware securing the right tail pipe to the tail pipe hanger. Resealed the tach cable where it passes through the firewall using black RTV silicone sealant. Replaced corroded Adel clamp on the scat duct under the #3 cylinder. Stop drilled 2 cracks in the engine baffle outboard of the alternator using a #30 drill bit. Replaced worn bolt in the left cowl flag hinge. Test ran the engine for 30 minutes with alternator under load. Adjusted the voltage regulator down to 28.5 volts at 2500 rpm. Replaced bad strobe bulb in the left wing tip. Matched PPG Paint to the original color of the pilot's plastic arm rest and applied the paint. Installed the plastic arm rest on the pilots interior side panel. Fuel selector fuel "off stop" button was hanging up on a placard Dressed the placard so the fuel selector fuel "off stop" button functions properly at this time. Tightened loose screw in the throttle vernier cable lock release button and checked tightness of remaining engine control cable lock release button screws. Replaced bushings in both of the aft nose gear door hinges. Replaced screws in the heat scat duct aft of the right back seat. Touched up paint on the roof antenna base and the front of the LED beacon adapter on the roof. Touched up blue paint on the rudder trim tab. Removed the tile down casting from the aircraft. Removed the internal tie down support by drilling out the rivets that were still intact and three additional broken rivets. Drilled the aft end of the tail cone bottom skin loose. Straightened the dented bottom skin. Installed tie down support back inside the tail cone using machine screws and nuts and riveted the straightened skin back in place. Installed the tail tie down casting back in place on the bottom of the tail cone using the original screws. Stop drilled the ends of two cracks in the aircraft belly skin. Removed ventilation duct work and other accessories from below the cabin floor to gain access to top side of the belly skin for riveting purposes. Installed a scab patch at fuselage station 86 just to the right of rectangular gear bellcrank cover. The scat patch was installed using MS20470AD rivets. Several rivets were drilled out of the center section bulkhead. The rivets used to replace those removed rivets and attach the scab patch to the belly skin were replaced using rivets of the same diameter, material and specifications as the rivets that were removed. The rivets were installed with appropriate tools in accordance with acceptable methods, techniques and practices called out in AC 43.13-1B chapter 4, paragraph 4-57 (a)(1, 2)(b, c), and figures 4-4, 4-5, 4-6, and table 4-8. Installed this scab patch in accordance with AC 43.13-1B Chapter 4, Section 4, "Metal Repair Procedures," paragraph 4-58 "Repair Methods and Precautions for aluminum structure", part (f) "Patching of Small Holes". This scap patch was patterned after typical patches on figure 4-16, titled "Typical Repairs of Stressed Sheet Metal.

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Covering." The rivet spacing for this repair was determined using the methods called out in AC 43.13-1B chapter 4 section 4 paragraph 4-57(b, c). Table 4-9. Reinstalled all ductwork and cabin accessories removed for this repair. Tightened both front bolts in the elevator trim tab push rods and loosened the aft bolt on the right trim tab push rod. Removed both of the brake rotors from the aircraft and installed new brake rotors, P/N 164-02505 and replaced all brake lining, P/N 066-10500. The brakes and rotors were conditioned in accordance with the instructions supplied by the brake lining manufacturer.

This aircraft has been inspected in accordance with an ANNUAL Inspection and was determined to be in AIRWORTHY condition.

Pertinent details of this inspection are on file at this agency under work order number 14-8390. This ANNUAL was completed

03/08/2014

Roger E. Smith

3442201 I.A.

Carter Aircraft, Inc. Sebring F

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
	Removed the volt new voltage regul Regulator Adjustr The aircraft and/o	Time: 169 age regulat ator per Bo nents. r componer Y for retur	or PN. 36-380 nanza 33,35, nt identified ab n to service. P	BEECH F33A; S/N: CE-1235; Tach: 1691.10; Total Time: 1691.10; Engine - Type: IO-526 ype: 3A32C406-C/G-82NDB-2, S/N: 871693, Time: 585.40 ype: 3A32C406-C/G-82NDB-2, S/N: 3A32C406-C/G-82NDB-2, S/N: 3A32C406-C/G-82NDB-2, S/N: 3A32C406-C/G-82NDB-2, S/N: 3A32C406-C/G-82NDB-2, S/N: 3A32C406-C/G-82NDB-2
	Removed the alternation break in the brushes new gasket. Tested elastomeric alternational alternator gasket. The Teledyne Continents and they both cover output. Inspected the The aircraft and/or continue the continue to the cover output.	ator from the Checked and install or coupler to some the IO-520 Parties IO-520 alternator omponent if or return to SMITH	e engine and field resistance of the alterna esting procedured in accounts Catalog Paeries of engine for oil leakage dentified above service. Per	ECH F33A; S/N: CE-1235; Tach: 1698.50; Total Time: 1698.50; Engine - Type: IO-520 of 3A32C406-C/G-82NDB-2, S/N: 871693, Time: 592.80  removed the elastomer coupler from the alternator. Spun the alternator on the bench to be and found it to be steady at 15 OHM. Installed the alternator on the engine using a stor coupler in accordance with TCM SB11-3, Dated 03/17/2010, which covers ure is in the Continental file. Installed the alternator on the engine using a new cordance with Teledyne Continental IO-520 Overhaul Manual P/N X30040A and P/N X30040A. Both the parts catalog and overhaul manual have the same part number ines. Test ran the engine and verified alternator operation including maximum voltage and no leakage was found.  The was repaired and inspected in accordance with current FAA regulations and was timent details of this repair are on file at this agency under Work Order Number 8434.  LUW4R585M.
Rem in the funct The : found 5/22/	oved the circuit boar e lead by soldering a lioned properly. aircraft and/or compo	from the cross the bonent identification to ser	dux terminal of teak. Installed lied above wa vice Platinen	A F33A; S/N: CE-1235; Tach: 1715.80; Total Time: 1715.80 on the alternator and found the circuit board has a broken lead. Repaired the break of the board on the alternator and test ran the engine. The alternator warning light as repaired and inspected in accordance with current FAA regulations and was not details of this repair are on file at this agency under Work Order Number 8454.  7. UW4R585M.

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Entries must be endorsed with rvalle, riding RECURUMO ) YEAR Technician or Repair Facility. (See back pages for other specific entries.) TIME IN FLIGHT TACH A 20 SERVICE TIME DATE Date: 4/27/2014; Aircraft: N1235C; Type: BEECH F33A; S/N: CE-1235; Tach: 1698.50; Total Time: 1698.50 Inspected all of the wiring on the alternator and the regulator. Removed the (FAST) regulator P/N 36-380096-1 and installed a used serviceable (SLOW) relay P/N 36-380056-5. Test ran the engine and verified that the regulator voltage is set correctly. The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found AIRWORTHY for return to service. Pertinent details of this repair are on file at this agency under Work Order Number 8437. JW4R585M. 4/27/2014 ROGER E. SMITH LATTER Carter Aircraft, Inc. Repair Station, Sebring FL. Date: 2/03/2014; Aircraft: N1235C; Type: BEECH F33A; S/N; CE-1235; Tach: 1659-30; Total Time: 1659-30 Removed the existing ELT Aftex Model ELT 110-4, S/N 68338, 125 MHZ ELT, antenna and the pilots manual control switch from the aircraft. Installed Artex ME405 ELT, P/N 453-6603 Rev B, S/N 07223, 406 MHz ELT kit, HEX ID ADCC4070DC 00261. This kit consists of the following components: ELT Transmitter w/ integral battery, Good Till July 2020, ELT Mounting Tray and Velcro Holder, ELT Antenna and Coaxial Cable Assembly, ELT Remote Switch and Remote Wiring Cable Assembly and ELT Audible Buzzer. This ELT system was installed in accordance with the Artex ME406 Series Emergency Locater Transmitter Manual P/N 570 -1600 Rev N dated June 25, 2013. Page 48 of this manual states that this Manual constitutes FAA approved data as described in AC 43.9-1E, Paragraph 6.h(2) and AC 43-210, Chapter 2, Paragraph 201(a)(6) and as such may be used as support for FAA Field Approval. This ELT was installed on the priginal ELT shelf provided by Beechcraft following the airframe manufacturers recommendations in the Beech 33, 35, 35 Maintenance Manual P/N 36-590001-9 which covers Model F33A S/N CE-748, CE-772 and after. This installation also meets the requirements of AC 43,13-2 " Acceptable Methods, Techniques, and Practices - Aircraft Alterations," specifically Chapters 1 thorough 3, 11, and 13. The aircraft owner will register this unit with NOAA using his personal data to complete the form. The aircraft equipment list was modified to reflect this alteration, Negligible change in weight and balance. The direcraft owner will register this beacon with NOAA per instructions in the installation manual for this equipment and instructions on the NOAA website. The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found AIRWORTHY for return to service Pertinent details of this repair are on file at this agency under Work Order Number 8328. UW4R585M. 2/03/2014 ROGER E. SMITH Carter Aircraft, Inc. Repair Station, Sebring FL.

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RECORDING Description of Inspections, Tests, Repairs and Alterations TODAY'S TOTAL YEAR TIME IN TACH FLIGHT Entries must be endorsed with Name, Rating and Certificate Number of SERVICE Technician or Repair Facility. (See back pages for other specific entries.) DATE: 9/11/2014; Aircraft: N1235C; Type: BEECH F33A; S/N. CE-1235; Tach: 1741.20; Total Time: 1741.20 Removed the front seats and inspected the landing gear up and down limit switches for condition and wiring security. Found no defect. Inspected the down limit switch and associated wiring on top of the landing gear actuator and no defects were noted. Checked all plugs in the system and found a loose plug under the left instrument panel for the electrical junction box. Separated plug # 354 from plug # P54 and found a corroded pin and socket. Cleaned the pin and socket with Electrical Grade Contact Cleaner and plugged the two plugs into each other. The transit light and all other landing gear position indicating lights functioned properly. Cycled the gear 6 times checking position indicating lights and gear warning system, all functioned properly. Removed the aircraft from the jacks and takied around on the ramp. The transition light never flickered. This work was performed in accordance with the Beech Model Bonanza (28V) Wiring Digram Manual P/N AF35-0102-9, diagram 32-60-01 Configuration 1 and the Beech 33, 35, 36 Maintenance Manual P/N 35-590001-9. Charged the battery with a Concord P/N 4142 battery charger in accordance with Gill Manual titled "Instructions, Activation, Charging and Continued Airworthiness", TBP 1560 Rev. H, dated 11/19/2008. Checked the The aircraft and/or component identified above was repaired and inspected in accordance with current FAA regulations and was found AIRWORTHY for return to service Pertinent details of this repair are on file at this agency under Work Order Number 8527. UW4R585M. Carter Aircraft, Inc. Repair Station. Settrino FI

# Switlik Aviation Maintenance, Inc.

Page Field Airport (FMY) 207 Danley Drive Fort Myers, Florida 33907

235-330-3000	WWW.SWILLINGVIAL	www.switlikaviation.com														
Reg. No.	S./N.	Date	Hobbs	T.T.	S.M.OH.	Page										
N1235C	CE-1235	5/13/2015	Tach 1787.2													
Serviced chara	and and installs	d = = = O:II OO 40	haddan Installad			all and a second										

Serviced, charged and installed new Gill G242 battery. Installed new fuel cap seals

ption of mapections, rests, Repairs and Alteration Entries must be endorsed with Name, Rating and Certificate Number TACH FLIGHT TIME IN Technician or Repair Facility. (See back pages for other specific entri TIME SERVICE DATE Stitch Works N1235C July 17, 2015 Tach: 1,803.0 Inspected interior installation after it had been refurbished and reinstalled by Stitch Works of Olathe, Kansas. Interior components refurbished at this time were all seats, headliner and arm rest. All materials used meet or exceeded requirements of FAR 25.853. Complied with Beechcraft AD 97-14-15, check proper locking and rotation of forward and aft door handles. Burn certificated provided to owner by Stitch Works of Olathe, Kansas. W&B change negligible. All work done in accordance with Beechcraft Maintenance Manual and AC 43.13-1B. Aircraft approved for return to service for work completed. Thomas A. Stellwag II AP2820575 Thomas G. Stilly 21

Description of Inspections, Tests, Repairs and Alterations TACH FLIGHT TIME IN Entries must be endorsed with Name, Rating and Certificate Number of TIME SERVICE Technician or Repair Facility. (See back pages for other specific entries.) Airframe Log Date 11/25/2015 Tach 1821.5 REMOVE MCCAULEY M/N: 3A32C406-6 S/N: 871693 PROPELLER (-63) LBS AND MCCAULEY P/N: D5259/D5541 SPINNER (-5.0) LBS. BOTH AT AN ARM OF 1" AND INSTALL NEW HARTZELL M/N: PHC-C3YF-1RF/F8068 S/N: QC1081B PROPELLER. (+75.0) LBS. AND P/N: A-2295-1/-2 (P) SPINNER +4.5 LBS. BOTH AT AN ARM OF 1". INSTALLATION DONE IN ACCORDANCE WITH STC #SA10135SC AND HARTZELL INSTALLATION INSTRUCTIONS. SEE 337 DATED 11/20/2015 FOR DETAILSREMOVE P/N: PAI-700 S/N: 76722 PRECISION AVIATION VERTICAL CARD MAGNETIC COMPASS FROM INSTRUMENT PANNEL GLARE SHIELD AND INSTALL P/N: C2400L4N S/N: 11152 AIRPATH COMPASS, OVERHAULED BY TIFFIN AIRE INC SEE WORK ORDER #3866 DATED 11/20/2015 FOR DETAILS OF OVERHAUL. REPLACE ALL BRAKE SYSTEM FLEXIBLE HOSE SEE INVOICE #593520 DATED 11/09/2015 FROM PHT INC. FOR ORIGIN OF HOSE. BLEAD BRAKES AND SERVICE RESERVOIR WITH 5606 FLUID. AIRCRAFT WEIGHED ON EVERGREEN DIGITAL SCALES M/N: 3-00057-BA-3C/B15B6B2 CALIBRATION DATE OF 11/18/2015. AIRCRAFT WEIGHED WITH FULL FUEL AND OIL. SEE WEIGHT AND BALANCE REVISION DATED 11/25/2015 FROM TIFFIN AIRE INC. FOR DETAILS. NEW EMPTY WEIGHT 2387.0 LBS, NEW EMPTY WEIGHT CG 82.54, NEW MOMENT 197022.00, NEW USEFULL LOAD 1013.00. Work performed by Tiffin Aire, Inc. Signature Thomas C.C. La A/P2753912IA



N1235C

SN: CE-1235

Tach: 1827.7

"I certify that the altimeters and static system tests and inspection required by FAR Part 91.411 have been performed IAW the requirements stated per paragraphs (a) and (b), Appendix E, FAR Part 43."

LH ADC Make: Garmin Model: GDC-74 S/N: 20607562 has been tested to 20,000 feet.

Standby Altimeter Make: United Model: 5934DA-1 S/N: 4E707 has been tested to 20,000 feet.

The Altitude Reporting system tested to 20,000 ft. per paragraph (c), Appendix E, FAR Part 43.

"I certify the ATC transponder tests and inspections required by FAR Part 91.413 were performed this date and found to comply with FAR Part 43, Appendix F."

#1 Make: Garmin Model: GTX-330 S/N: 84135125

The Maintenance Identified In This Work Order Package Has Been Inspected And The Item(s) Are Approved For Return To Service. Pertinent Details Of The Repair(s) Are On File At This Agency Under Work Order No. DAY-WO-4271.

SIGNED Tel Water

For STEVENS AVIATION, INC., DAYTON, OH CRS VIB4368K

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ALTIMETE	FAR PART 43, DATE: 29/2016 S/N: 4E707 Standby	0001-	1500	2000	3000	4000	0009	0000	0000	ALTIMETER FAR PART 4	DATE: 2/9/2016 S/N: 20607562 Pilot	-1000	0	200	1500	2000	3000	4000	0009	8000	10000	12000

A N

TIRES PACK REAS ADJU AFTE

36-38 DUE BATT CABL

REPL

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Work

N1235C Airframe Log Date 04/21/2016 Total Time 1841.2 Tach 1841.2 ANNUAL INSPECTION COMPLETED THIS DATE IN ACCORDANCE WITH BEECH INSPECTION GUIDE 36-590001-9: REMOVED TIRES, CLEANED, MEDIA BLASTED, ETCHED, PRIMED AND PRINTED ALL WHEEL HALVES AND BRAKE CASTING. REASSEMBLED, PACKED ALL WHEEL BEARINGS. REMOVED LEFT GEAR STRUT AND REPLACED ALL SEALS, WIPERS, AND O-RINGS, CLEANED, REASSEMBLED AND SERVICED WITH NITROGEN/OVER MIL SPEC 5606 HYDRAULIC FLUID. INSPECTED GEAR MOTOR BRUSHES. ADJUSTED NOSE GEAR DOORS. GREASED GEAR. PERFORMED RETRACTION AND EMERGENCY EXTENSIONS. BLEED BRAKES AFTER REBUILD WITH 4 EACH MS28775-222 O-RINGS AND REPLACING BOTH BRAKE MASTER CYLINDERS WITH 96-380034-25. REPLACED BOTH LH AND RH GEAR UP LOCK BOOTS 35-815156-5 AND 35-815156-4. REPLACED FUEL SELECTOR VALVE 36-380026-1 S/N 5887 AND TESTED OPERATION. ELT INSPECTED FOR CORROSION AND TESTED OPERATION ELT BATTERY DUE 7-20. CAP CHECKED SHIPS BATTERY AT 110.3% - TOP CHARGED AND INSTALLED IN NEWLY CLEANED AND PAINTED BATTERY BOX. RETORQUED HOSE CLAMPS ON FUEL TANK NIPPLES AND RESEALED FUEL VENTS. ADJUSTED ELEVATOR CABLE TENSION. REATTACHED DEFROST CONTROL CABLE TO DEFROST VALVE LEVER. REPOSITIONED LATCHING ANCHORS ON RIGHT EMERGENCY WINDOW. CLEANED AND LUBRICATED BOTH WINDOW SEALS. REPLACED ALL WING ATTACH BOLTS, NUTS, AND WASHERS AND LUBRICATED PER BEECH MAINTENANCE MANUAL. RUDDER FWD. SPAR DUE FOR INSPECTION AT TACH 2057.9 PER A.D. 93-23-03. A.D. 09-25-01 PILOT/CO-PILOT SHOULDER HARNESS RETAINING WASHERS FOUND TO BE WITHIN DEMENTIONS OF MSB25-2394 REV. 1. ALL APPLICABLE A.D.'S THROUGH 2016-08 COMPLIED WITH. Altimeter & Static Check Due: 2-18 ELT Battery Due: 7-20 ELT Operation and Corrosion Check Due: 4-17 Transponder Encoder Check Due: 2-18 Weight due: Empty Wt. 2387.00 C.G. 82.54 Useful 1013.00 Gross 3400 certify that this airframe has been inspected in accordance with an annual inspection and was determined to be in airworthy condition. IA3462159 Work performed by Tiffin Aire, Inc. Signature

\*\*\*\*\*\*MUNCIE AVIATION COMPANY C.R.S. # AJVR135C \*\*\*\*\*\* ations RECORDING YEAR Tach Make: Beech Ser. No.: CE-1235 YEAR Reg Mark Date iber of TACH 1844.3 Model: F33A Reg. Mark: N1235C N1235C 5/6/2016 20 TIME entries.) DATE The aircraft / component identified above was repaired and Removed and replaced Pilots PTT DATE inspected in accordance with current Regulations of the switch. Ramp checks good. Federal Aviation Administration and found airworthy, and is approved for return to service for work performed. Pertinent details of the repair are on file at the Agency. Work Order No.: A004392 Date: 05-06-2016 Inspector for Signed: \_\_\_ MUNCIE AVIATION COMPANY C.R.S.#AJVR135C MUNCIE, IN 47303-9778 (765) 289-7141 \*MUNCIE AVIATION COMPANY C.R.S. # AJVR135C\* Tach Make: Beech Ser No.: CE-1235 Reg Mark Date 1854 Model F33A Reg. Mark: N1235C N1235C 6/27/2016 The aircraft / component identified above was repaired and See Revised Weight and Balance/ Equipment list dated this date for a detailed inspected in accordance with current Regulations of the list of all removed and installed equipment. Federal Aviation Administration and found airworthy, and Installed GDL88 P/N 011-02369-00 S/N is approved for return to service for work performed. 22T004925 IAW STC # SA02119SE. Pertinent details of the repair are on file at the Agency. Installed ESI500 P/N 9200-15000-0101 Work Order No.: \_\_A004185 Date: \_06-27-2016 S/N 1022 IAW STC # SA04046CH. Installed Signed: Inspector for G1 Engine Monitor System IAW STC # MUNCIE AVIATION COMPANY C.R.S.#AJVR135C SA02930NY. Updated G500 Software to MUNCIE, IN 47303-9778 (765) 289-7141 Ver. 7:00 IAW Garmin STC SB 1524 Rev. A, and applied STC # SA02153LA-D to the G500 installation IAW SB. See FAA From 337 dated this date for details. Installed CI-105-16 GDL88 Antenna S/N 35775 with doubler IAW AC43.13-2B Chapter 3. Installed KX165 NAV/COM P/N 069-1025-25 S/N 48167 IAW Installation Manual 155-01345-0000 Rev. 4. Installed CO Detector P/N 553-101-00 S/N 50322 IAW Installation Manual 01-553-01. Updated GNS530W Software to Ver. 5.30 IAW STC SB 1618 Rev. A. Removed Vacuum Systems from Aircraft along with all hoses and fittings, placed blank off plate on engine vacuum mount. Pitot and static systems leak check good. The transponder has been tested as required by 91 413 and was found to comply with the limits set in FAR 43 appendix F. The installed ADS-B system was shown to meet the equipment performance requirements of 14 CFR section 91 227 All systems mentioned in this lookook entry ramp chacks

MFG. DATE (MM/DD/YY): 06/21/16 PART # 8102 REV COUNTRY United States (366) A/C #(HEX) N/A A/C #(OCT) N/A BLOCK # N/A A/C TAIL #: N/A 15-DIGIT HEX ID#:2DC88 59636 FFBFF

> ACR ELECTRONICS, Inc. DATE BOOK ENTRY CEMENT E.L.T.

CAGE Code 18560

# \*\*\*\*\*\*MUNCIE AVIATION COMPANY C.R.S. # AJVR135C

Tach Make: Beech Ser. No.: CE-1235 Reg Mark Date N1235C 6/27/2016 Removed KC290 Autopilot Controller S/N 11138 and KA285 S/N 30109, repaired. bench checked good IAW KFC200 Autopilot Acc. Manual 006-05556-0005. and reinstalled, Removed KS270A Pitch Servo S/N 22531, repaired, bench checked good IAW Maint Manual 006-05288-0004. and reinstalled. Removed KC295

Autopilot Computer S/N 10948, repaired,

1854 Model: F33A Reg Mark: N1235C The aircraft / component identified above was repaired and inspected in accordance with current Regulations of the Federal Aviation Administration and found airworthy, and is approved for return to service for work performed. Pertinent details of the repair are on file at the Agency. Work Order No.: A004185 Date: 06-27-2016 Signed Inspector for MUNCIE AVIATION COMPANY C.R.S.#AJVR135C MUNCIE, IN 47303-9778 (765) 289-7141

bench checked good IAW KC295 Maint. Manual 006-05558-0003, and reinstalled. Autopilot ramp checks good

# MUNCIE AVIATION COMPANY C.R.S. # AJVR135C

Reg. Mark N1235C 8/8/2016 Removed Artex ME406 ELT P/N 453-6603 Switch P/N 345-6196-04, Installed Artex 345 System IAW Artex 345 Installation Manual Y1-03-0282 Rev. C. ELT 345 P/N A3-06-2880 S/N 267-02843, ELT Remote Switch P/N A3-06-2759 S/N 370889-044. and ELT Antenna P/N A3-06-2892-1 S/N 370847-005. All new equipment installed

Tach Make: Beech Ser No.: CE-1235 1863.51854 Model: F33A Reg Mark: N1235C

The aircraft / component identified above was repaired and S/N 07223, ELT Antenna, and ELT Remote inspected in accordance with current Regulations of the Federal Aviation Administration and found airworthy, and is approved for return to service for work performed. Pertinent details of the repair are on file at the Agency. Work Order No.: \_\_A004498\_Date: \_08-08-2016 Signed Inspector for MUNCIE AVIACION COMPANY C.R.S.#AJVR135C MUNCIE, IN 47303-9778 (765) 289-7141

in same location as old equipment. Weight change is negligible, no change to aircraft weight and balance. Removed KX165 S/N 48167, bench checked good IAW KX155/165 Maint Manual 006-05179-0009, and reinstalled. All systems mentioned in this logbook entry ramp checks good.

YEAR 20 DATE	RECORDING TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
	Reg. M N12350 Swappi warning Probe v Remov Detecto	Mark Date C 9/ ed position of g lights. Remove the P/N 102-led and replace or OAT Probe	1/2016 clock and Gear oved and replace 00007 S/N 4607 ced Guardian CO with P/N AD590 ed in this logboo	red CHT inspected in accordance with current Regulations of the Federal Aviation Administration and found airworthy, and is approved for return to service for work performed. Pertinent details of the repairer on file at the Agency.

#### AIRFRAME LOG

DATE: 4/21//2017

REGISTRATION NUMBER: N1235C

MAKE: RATHEON MODEL: F33A SERIAL: CE1235

Tach: 1886.9 ACTT: 1886.9

20 DA

- Completed an annual inspection in accordance with FAR 91.409 using the Textron F33A maintenance manual as a guide and this aircraft has been determined to be airworthy.
- Rigged the cabin air vent control.
- Reinstalled the pilot's armrest.
- Repaired the wiring for the wing strobe lights. Tested good.
- Serviced the main aircraft battery.
- Completed the 12 month portable fire extinguisher inspection.
- Replaced the fuel cap seals with new.
- Complied with FAR 91.207d by inspection and test of the ELT. No defects noted. Due again in 12 months.
- Complied with AD 84-26-04 by replacement of the induction air filter with a new P13-7627 filter. Due again in 500 hours
- All above described work completed in accordance with the Textron F33A maintenance manual.
- 11. AD's verified thru current issue.
- 12 Ground run and checked good.

THE MAINTENANCE PERFORMED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH FAA REGULATIONS AND APPROVED FOR RETURN TO SERVICE

ge Heggs AP344477/ILA

20 DATE	TACH	FLIGHT	TIME IN SERVICE	Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
		Reg. Mark Date N1235C 1 Removed GA35 0 013-00235-00 S/I 140629 IAW ICA Updated G500 Sc Garmin STC SB 1 placed in POH. C ESI500 Battery IA Manual P/N 0040	PS/WAAS Antenna SPS/WAAS Antenna N 55099 and installed P/N 005-00638-09 of oftware to Ver. 7.22 1735 Rev. A. AFMS alibrated and charg W ESI500 Installation	led S/N Rev. 3. Federal Aviation Administration and found airworthy, and is approved for return to service for work performed.  Pertinent details of the repair are on file at the Agency. Work Order No.: A006185 Date: 11-20-2017  Signed: Inspector for MUNCIE AVIATION COMPANY C.R.S.#AJVR135C

N1235C

SERVICE

SN: CE-1235

Technician or Renair Facility (See back names for other consilio cation

"I certify that the altimeters and static system tests and inspection required by FAR Part 91.411 have been performed IAW the requirements stated per paragraphs (a) and (b), Appendix E, FAR Part 43."

LH ADC Make: Garmin Model: GDC-74A S/N: 20607562 has been tested to 20,000 feet. RH ADC Make: L-3 Model: ESI 500 S/N: 1099 has been tested to 20,000 feet. Stby Alt Make: United Model: 5934AD-1 S/N 4E707 has been tested to 20,000 feet.

The Altitude Reporting system tested to 20,000 ft. per paragraph (c), Appendix E, FAR Part 43. "I certify the ATC transponder tests and inspections required by FAR Part 91.413 were performed this date and found to comply with FAR Part 43, Appendix F."

#1 Make: Garmin Model: GTX 330 S/N: 84135125

The Maintenance Identified In This Work Order Package Has Been Inspected And The Item(s) Are Approved For Return To Service. Pertinent Details Of The Repair(s) Are On File At This Agency Under Work Order No. DAY-WO-5198.

SIGNED

For STEVENS AVIATION, INC., DAYTON, OH CRS VIB4368K

Description or inspections, Tests, Repairs and Alterations TACH FLIGHT TIME IN Entries must be endorsed with Name, Rating and Certificate Number of TIME Technician or Repair Facility. (See back pages for other specific entries.) SERVICE DATE STEVENS ALTIMETER TESTED & INSPECTED PER ALTIMETER TESTED & INSPECTED PER ALTIMETER TESTED & INSPECTED PER FAR PART 43, SECT. 91 411 FAR PART 43, SECT. 91 411 FAR PART 43, SECT. 91.411 DATE: 3/27/2018 DATE: 3/27/2018 DATE: 3/27/2018 S/N 20607562 S/N: 4E707 S/N: 1099 Pilot Copilot Stndby -1000 -10 -1000 -10 -1000 -10 +20 -5 -5 -10 +25 -5 -5 -10 +25 -10 -10 -5 +35 -5 -5 -5 +15 +15 +20 

Appering Coll Armine Advert County FRA/Doiled States	AUTHORI	ZED RELEAS	E CEI	RTIFICATE	E. Wars O	5932343	
Chinaman Name and Address	Textron A	viation Inc. (PC4) n Aviation Inc. d, Wichita, KS 67:			Number	2933711	
20 TIRE 5.00 FLC	X5 6PLY	505C66-5	1	701321	)2	New	6
1 to Constant the term identified  Approved design data on  Sequesproved design de  136 Authorized Signature	dure in a condition for sale of	13c. Append d Audionia	Carolina Br 11 and these Code of the applicated to	at Victory of feeder 12 Of of address offerware specified as possed to finish TI was accommod from Propolations, part TI and a memory to many year.	THE RES	cont identified to Dioch control and Title 14. Self week the state for Application, Waster 140.	Textron F33A maintenance manual noted. Due again in 12 months. its noted. nd inspected. Assembled with new
The Name (Typed or Printed)  JOHNNY GLOV		13/Dec/2	2017	SAS Name (Typod to Yold		Design (A) Simple Syrty	nce manual. pected. Assembled with new seals anual.
Where the oper installer persons in Black 1, it is expended that the country specified to Block 1.	n work in accordance with it cuser morallor enough that h	nt ainte does not automatically of e rational regulations of an airwo e has airworthstane authority acco	ethiness mill opis singraft o	hereby to exceed the sourcest ong hereby different than the mirwest engint(s) propeller(s) article(s)			Assembled with a new 505066-5 rings. Serviced and installed the
Streemen is Stocks 13s and I- streetimes with the selfemal or FAA Form \$136.3 (02-14)	In the most committees (materialism polarisms by the ower (postation	m certification, In all cown, cerci- terfine the second may be flown				NSN 003248-012400	aintenance manual.

AND APPROVED FOR RETURN TO SERVICE

rof ies.)

Bills E. Higgs AF344477/IA

## Aviation Sales, Inc.

501 North Dixie Drive Vandalia, OH 45377

AIRFRAME LOG

DATE 4/02/2018 REGISTRATION NUMBER: N1235C

MAKE RATHEON MODEL: F33A SERIAL: CE1236 Tach: 1927.6 ACTT: 1927.6 YEAR

DATE

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les.)

Completed an annual inspection in accordance with FAR 91.409 using the Textron F33A maintenance manual
as a guide and this aircraft has been determined to be airworthy.

2. Complied with FAR 91.207d by inspection and test of the ELT. No defects noted. Due again in 12 months.

3. Complied with the 12 month portable fire extinguisher inspection. No defects noted.

 Removed the right main landing gear lower strut assembly. Disassembled and inspected. Assembled with new seals. Serviced and installed in accordance with the Textron F33A maintenance manual.

Removed the nose landing gear lower strut assembly. Disassembled and inspected. Assembled with new seels.
 Serviced and installed in accordance with the Textron F33A maintenance manual.

6. Repaired the left nose landing gear bonding strap as required.

Removed the nose wheel assembly. Disassembled and inspected the wheel. Assembled with a new 505C66-5
tire, SN: 70132102. Serviced and installed new LM67010 and LM67048 bearings. Serviced and installed the
assembly in accordance with the Textron F33A maintenance manual.

8. All above described work completed in accordance with the Textron F33A maintenance manual.

9. AD's verified thru current issue.

10. Ground run and checked good.

THE MAINTENANCE PERFORMED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH FAA REGULATIONS AND APPROVED FOR RETURN TO SERVICE.

ELECTING G. APTYYYTTIA

RECORDING TACH TIME

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### AIRFRAME I

DATE: 10/23/2018 REGISTRATION NUMB MAKE: RATHEON MODEL: F33A SERIAL: CE1235

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- 2. Rep
- 3. Gro

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4. Organismos in the contract of the contract	ASPORT Canada  Sizations Name and Address CHT INSTRUMENT CORPORATION SIDUSTRIAL DRIVE		ELEASE CERTII ORM ONE	FICATE	1 Form Tracking Nn. 130619 - 22
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NONE	b			166683	NEW
X Q	Approved design data and are in a cond	tition for safe operation.	D one regular	faintenpes Riferer  our specifies in block 17  5135 mins specifies in block 12.	Se week to maintain to be a second
C. Satisfaction	Approved design data and are in a cond Non-approved design data specified in the second secon	block 12.  Approved Organization Number  4 42-90	D one regular		Se wat a separtified to about 11 and account with the Consular of page 1.  Appropriate Organization, Nambur
A Manne	Approved design data and are in a cond Non-approved design data specified in the second secon	block 12.  Approved Organization Number  4 42-90	Other regular desilies the males described in the said regularisms.		Special acception in 1982 II and unconflict for Consider As in an in Appropriate Organization National Section Section 1987

# Aviation Sales, Inc.

501 North Dixie Drive Vandalia, OH 45377

## AIRFRAME LOG

DATE: 10/23/2018

REGISTRATION NUMBER: N1235C

MAKE: RATHEON MODEL: F33A SERIAL: CE1235

Tach: 1950.9 ACTT: 1950.9

- Removed the CO detector PN: 553-101-001, SN: 50322 for repair. Secured wiring. Weight change negligible.
- Replaced cylinder #6 EGT probe with a new 2870 probe.

THE MAINTENANCE PERFORMED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH FAA REGULATIONS Ground run and checked good.

AND APPROVED FOR RETURN TO SERVICE.

Liege 7699 A 17444771

YEAR	TACH TIME	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
				Aviation Sales, Inc. 501 North Dixie Drive Vandalia, OH 45377
	DATE: 12/19/2 REGISTRATIO MAKE: RATHE MODEL: F33A SERIAL: CE12	018 N NUMBER EON		Tach: 1953.9 ACTT: 1953.9
		Install     Replace	ed overhauled ced alternator d run and ched	CO detector PN: 553-101-001, SN: 50322 . Weight change negligible. out sensor with new P/N A-00258-2. oked good.
			PERFORME R RETURN T	ED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH FAA REGULATIONS TO SERVICE.
	<u> 7</u>	Elly E.	Hig G	7-AP3444771

N1235C SN: CE-1235  2/13/2019 HM: 1954.8  Removed Garmin GDU 620, p/n: 011-01264-50, s/n: 165202045. Installed an overhauled Garmin GDU 620, p/n: 011-01264-50, s/n: 165200279. GDU 620 ops checked good. Work performed IAW Garmin G600/G500 Maintenance Manual.  The Maintenance Identified In This Work Order Package Has Been Inspected And The Item(s) Are Approved For Return To Service. Pertinent Details Of The Repair(s) Are On File At This Agency Under Work Order No. DAY-WO-98.  SIGNED For STEVENS AEROSPACE LLC, DAYTON, OH CRS VIB4368K	ATE	TACH	TODAY'S FLIGHT	TOTAL TIME IN SERVICE	Description of Inspections, Tests, Repairs and Alterations Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)
O11-01264-50, s/n: 165200279. GDU 620 ops checked good. Work performed IAW Garmin G600/G500 Maintenance Manual.  The Maintenance Identified In This Work Order Package Has Been Inspected And The Item(s) Are Approved For Return To Service. Pertinent Details Of The Repair(s) Are On File At This Agency Under Work Order No. DAY-WO-98.	2/13/20	119			The state of the s
	The Ma Approv	ed For Return Work Order N	ntified In To Service o. DAY-W	ce. Pertinent	t Details Of The Repair(s) Are On File At This Agency

RECORDING TACH TIME IN SERVICE

Description of Inspections, Tests, Repairs and Alterations
Entries must be endorsed with Name, Rating and Certificate Number of Technician or Repair Facility. (See back pages for other specific entries.)

Aviation Sales, Inc.

501 North Dixie Drive Vandalia, OH 45377

### AIRFRAME LOG

DATE: 02/14/2019 REGISTRATION NUMBER: N1235C

MAKE: RATHEON MODEL: F33A SERIAL: CE1235 Tach: 1954.9 ACTT: 1954.9

- REPLACED BATTERY WITH NEW G242 S/N G03017917.
- Ground run and checked good.

THE MAINTENANCE PERFORMED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH FAA REGULATIONS AND APPROVED FOR RETURN TO SERVICE.

Belge. 2499-11

# Aviation Sales, Inc.

10600 Springboro Pike Miamisburg, OH 45342



r of ries.)

N1235C HM: 1957.7

05/13/2019 ACTT: 1957.7

Completed an annual inspection in accordance with FAR 91.409 using the Raytheon F33A maintenance manual as a guide and this aircraft has been determined to be airworthy. Complied with FAR 91.207d by inspection and test of the ELT. No defects noted. Due again in 12 months. Replaced all of the brake linings with new RA066-10500 linings. Secured and adjusted the cabin door upper latch assembly. Replaced the static discharger on the right elevator with a new discharger. Replaced the fuel cap seals with new seals. Removed the pilot and co-pilot seat belt assemblies. Assemblies repaired and rewebbed by C&M Marine and reinstalled. Removed the factory fuel quantity gauges. Installed an Aerospace Logic, Inc. FL202, SN: ASL009477 fuel quantity indication system in accordance with STC # SA02825NY. (See FAA337 this date.) Installed a RC Allen RCA2610-P2, SN: 919D239 electric standby attitude indicator in the center instrument panel in accordance with the Kelly Manufacturing Co. KMC Publication No. 1401-6. Leak checked pitot and static systems. 10. AD's verified thru current issue. II. All above described work completed in accordance with the Raytheon F33A maintenance manual. 12 Run up and checked good.

I certify that the above described work has been completed in accordance with current FAA regulations and the aircraft and/or appliance is approved for return to service.

gulallett 2396

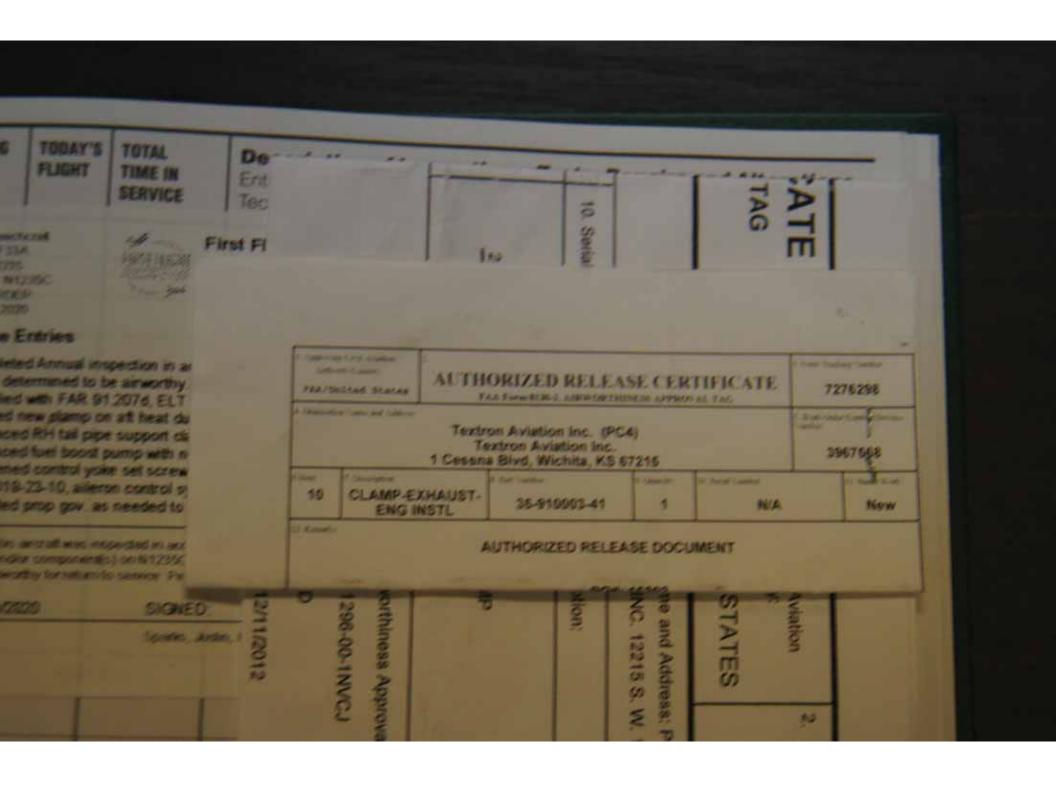
ins First Flight Aviation - CRS-GA5R-876M

10570 Springboro Pike Miamisburg. OH 45342 Phone 937-885-5580 FAX 937-885-5586 YEAR YEAR! of 20 Altimeter(s) tested & inspected per FAR 43, Appendix E. DATE DATE Stby: 5934AD-1 #2: ESI-500 #1: GDC 74 S/N: 4E707 S/N: 1099 S/N: 20607562 -1.000 -20 -1.0000 -10 -1,000 -10 0 -10 0 -10 -10 500 -10 500 -10 500 -5 1,000 -10 1,000 -10 1,000 -5 1,500 -10 1,500 -10 1,500 -5 2,000 -10 2,000 -10 2,000 -5 3.000 -10 3,000 3,000 -10 -5 4,000 -10 4,000 -10 4,000 0 6,000 -20 6,000 -20 6.000 10 8,000 -20 8,000 -20 8.000 20 10,000 -10 10,000 -20 10.000 Model: GTX 330 ES S/N. 84135125 Garmin Make: Details of this inspection are on file at this repair station under 6-10-2020 WO# 31725-06-2020 Date: Rya Cht Signed:

# TIPST FIIGHT A 10570 Springboro Pike Miamisburg, OH 45342 Phone 937-885-5580 FAA

"I certify that the altimeter and static tests and inspection required by FAR 91.411 have been performed. The altimeter(s) per para (A) and (B) appendix E FAR 43."

been performe	ed. The altimeter	Model:	A) and (B) appendi	_S/N:_	
	Garmin		ESI-500	S/N:	1099
ADC #2:_	L3	_ Model: _			4E707
	United	Model:	5934AD-1	S/N:_	
Stby Make: _ Has been te	-todto:	20,000	feet.		feet, per para (C)
annondiy F	FAR 43.		tion r	equired b	by FAR 91.413
appendix E "I certify the were perfor	FAR 43. e ATC transpormed this date  Carmin	and found to Mod are on file at	and inspection re	ES S/	by FAR 91.413 opendix F." N: <b>84135125</b>



MAKE Beechcraft MODEL F33A SN CE 1235 **REG NO N1235C** WORK ORDER 31704-06-2020



#### First Flight Aviation

10600 Springboro Pike Miamisburg, Ohio 45342 USA Phone 937-885-5580

DATE 6/15/2020 A/C TSN: 19734 TACH 19734

#### Airframe Entries

(2) Completed Annual inspection in accordance with FAR 91.409, referencing the Textron F33A maintenance manual. This aircraft has been determined to be airworthy. AD's researched thru current date.

(8) Complied with FAR 91.207d, ELT insp. Ops check normal. Battery due SEP/2023.

(9) Installed new clamp on aft heat duct.

(17) Replaced RH tail pipe support clamp with new 35-910003-41. (18) Replaced fuel boost pump with new p/n 1298-00-1NVCJ, s/n 288. P/n 1296-00-1NVCJ pd

(19) Tightened control yoke set screws.

(20) AD 2019-23-10, aileron control system, does not apply to this aircraft serial number.

(23) Adjusted prop gov. as needed to bring max RPM down.

and the second property of the second propert	exaction and found to be airworthy
I certify that this aircraft was inspected in accordance with an annual in	specially and loans to be all the Federal Austral Dominions and
HIL MADRE WAS FROMFORD AND INCOME.	ad in accordance with cliften takes tilles of the rederd aviduoli regulations on
Detined details of the report	are on tile at this report station under Work Order No. 31704-06-2020
was found Anworthy for return to service. Pertinent details of the repair	are on file at this repetir station under Work Order No. 31704-06-2020

DATE: 6/15/2020

SIGNED:

Sparks, Justin, IA 3442636

Work Order: 31704-06-2020

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