

1. Approving National Aviation Authority/County: FAA/UNITED STATES	AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. System Tracking Ref. No. S O # 623857
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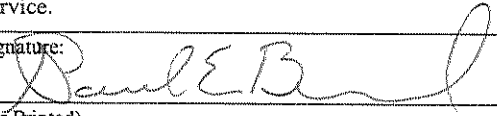
4. Organization: GARMIN International 1200 E 151st Olathe, KS 66062	5. Work Order Contract/Invoice Number: S O # 623857
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6. Item:	7. Description:	8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1.	GDL69A	011-00987-00	N/A	1	47750324	MODIFIED

13. REMARKS: Updated software. Operating software for this unit must be reloaded using the approved aircraft system software once the unit is reinstalled in the aircraft in order to complete the return to service process. Correct loading of this software can be verified by referring to the appropriate avionics/line maintenance manual.

This "Certifies that the work specified in block 12/13 was carried out in accordance with EASA Part-145 and in respect to that work the component is ready for release to service under EASA Part-145 Approval Number: EASA.145.5534". The work that was performed on this unit was done to meet the requirements of the Maintenance Manual, part number 190-00364-00.

14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation <input type="checkbox"/> Non-approved design data specified in block 13	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulations specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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15. Authorized Signature <div style="text-align: center;">N / A</div>	16. Approval/Authorization No.: <div style="text-align: center;">N / A</div>	20. Authorized Signature: 	21. Approval/Certificate No.: <div style="text-align: center;">G6XR582Y</div>
17. Name (Typed or Printed) <div style="text-align: center;">N / A</div>	18. Date (m/d/y) <div style="text-align: center;">N / A</div>	22. Name (Typed or Printed) <div style="text-align: center;">Paul E Brand</div>	23. Date (m/d/y) <div style="text-align: center;">11/20/2007</div>


User/Installer Responsibilities

It is important to understand that the existence of this Document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block 1.

Statements in block 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

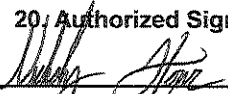
1. Approval National Aviation Authority/Country: FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: WRK0121823
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4. Organization Name and Address:  MID-CONTINENT INSTRUMENT CO., INC 9400 E 34th ST. NORTH WICHITA, KS 67226-2615 FAA CRS OL2R061L	5. Work Order/Contract/Invoice Number: WRK0121823
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6. Item:	7. Description:	8. Part Number:	9. Eligibility:	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	ATTITUDE GYRO	4300-205 4300	N/A	1	F05-11956	OVERHAULED

13. Remarks:
THE WORK ORDER REFERENCED DESCRIBES THE ACTUAL WORK PERFORMED.
ALL WORK WAS PERFORMED REF. MANUAL # MCI SPEC. 4300-205 REV. B, 5/11/05
 The work specified in block 12/13 except as otherwise specified was carried out in accordance with EASA Part 145 and in respect to that work the aircraft component is considered ready for release to service under EASA Approval Certificate Number EASA.145.4675.

14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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15. Authorized Signature:	16. Approval/Authorization No.:	20. Authorized Signature 	21. Approval/Certificate No.: FAA CRS OL2R061L
17. Name (Typed or Printed):	18. Date (m/d/y):	22. Name (Typed or Printed): NICHOLAS HOWE	23. Date (m/d/y): May/12/2008

User/Installer Responsibilities

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Final Work Order

omenclature ATTITUDE GYRO	Serial No F05-11956	Location KS	RMA # RMA000000142260	AR # 3310
Manufacturer's Part No. 4300-205	Model No. 4300		Quantity 1	
Manufacturer MCI	Customer Item Number		Service Period	
Customer HAWKER BEECHCRAFT CORP - RAPID	Customer PO No. 4502605373		Customer Reference No.	

9015645	POT 1K SMD	1
9016135	RETAINING RING	2
RI-1438ZZSD	BEARING	1



April 29, 2008

Page: 1

Final Work Order

Designation ATTITUDE GYRO	Serial No F05-11956	Location KS	RMA # RMA000000142260	AR # 3310
Manufacturer's Part No. 4300-205	Model No. 4300		Quantity 1	
Manufacturer MCI	Customer Item Number		Service Period	
Customer HAWKER BEECHCRAFT CORP - RAPID	Customer PO No. 4502605373		Customer Reference No.	

Reason For Removal
 HOLE ON SIDE.

Work Instructions Panel Tilt: 7.5° Input: 10-32V LTG: 28V
 Warranty Repair _____ Overhaul Return As Is _____ Special Instructions
 Functional Test _____ Repair _____ Modified _____

Preliminary Inspection
 Seals Broken No Cover Damaged Yes Hidden Damage Inspection Required No
 Nameplate Damaged No Finding HOLE IN CASE.

Preliminary Inspector KELLEY ROSEBAUGH

Date April 29, 2008

Failure Analysis

Customer Complaint Verified

TAIL SHAFT BEARINGS ARE ROUGH, ROTOR IS NOISY, NEEDS REV D.

Previous Service Bulletins Installed: MODS 1-3.

Work Accomplished

DISASSEMBLED, CLEANED, INSPECTED, REPAIRED AS NECESSARY USING PARTS LISTED, REASSEMBLED AND INSPECTED IN ACCORDANCE WITH MANUFACTURERS SPECIFICATIONS. UNIT IS OVERHAULED. INSTALLED REV D.

I.A.W Manual # MCI SPEC. 4300-205 REV. B, 5/11/05

Outgoing Service Bulletins MODS 1-3.

Technician ROBERT BIGGARS

Date May 10, 2008

In-Process Inspector JOE RUIZ

Date May 12, 2008


FINAL INSPECTOR NICHOLAS HOWE *Nicholas Howe*

Date May 12, 2008

Parts Used

Part Number	Part Description	Quantity
53-6800J1TK-E	RESISTOR 680	1
9010075-1	HOUSING, BEARIN	2
9015359	ROTOR WHEEL	1
9015365	ENDCAP TOP	1
9015545	COVER, EGH	1
9015607	STANDBY BATTERY	1
MZ60160X48A7	BEARING	2
SSR4ZZMCRA7P24	BEARING, BALL	1

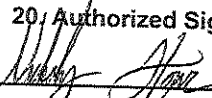
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17. Name (Typed or Printed):	18. Date (m/d/y):	22. Name (Typed or Printed): NICHOLAS HOWE	23. Date (m/d/y): May/12/2008

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**INSTRUMENT, STATIC AND AVIONICS CHECKS
RECORD OF COMPLIANCE**

Aircraft Model G-36 Serial Number E-3730
 Reg # N324P Owner's Name _____

FAMEMON AIRCRAFT INSTRUMENTS
 8001 LEMMON AVE.
 DALLAS, TX 75209
 FAA CRS# RN1R458K

THE ALTIMETER SYSTEM, ALTITUDE REPORTING, PITOT/STATIC SYSTEM LEAK TESTS AND INSPECTIONS REQUIRED BY 14 CFR PART 91.411 HAVE BEEN PERFORMED AND FOUND TO COMPLY WITH APPENDIX E, PART 43, FAA REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE.

MAX. ALTITUDE: 20,000 FT.

#1 A.D.C. P/N: 011-0082-00 S/N: 47807577 (WO# 14221)
 #2 ALT. P/N: 5934-3 S/N: 459475 (WO# 14222)
 TYPE AIRCRAFT: BOANAZA G-36 S/N: E-3730
 REGISTRATION: N324P
 DATE: 20-FEB-2009

SIGNATURE *[Handwritten Signature]*

COMPLYING AGENCY	NOTATIONS	NEXT DUE

ALTIMETER SCALE #1 P/N: 011-0082-00
 CORRECTION CARD N324P S/N: 47807577

STANDARD ALTITUDE FT.	ALTIMETER READING	STANDARD ALTITUDE FT.	ALTIMETER READING
-1000	-5	16000	+5
0	+5	18000	+5
500	-5	20000	+5
1000	-5	22000	

ALTIMETER SCALE ST-BY P/N: 5934-3
 CORRECTION CARD N324P S/N: 459475

STANDARD ALTITUDE FT.	ALTIMETER READING	STANDARD ALTITUDE FT.	ALTIMETER READING
-1000	+20	16000	-40
0	-5	18000	-35
500	-15	20000	-30
1000	-10	22000	
1500	-15	25000	
2000	-10	30000	
3000	-5	35000	
4000	-5	40000	
6000	-15	45000	
8000	-25	50000	
10000	-30	51000	
12000	-40	52000	
14000	-45	53000	

DATE: 2-20-09 WORK ORDER # 14222

MEMON AIRCRAFT INSTRUMENTS

INSTRUMENT, STATIC AND AVIONICS CHECKS RECORD OF COMPLIANCE

Aircraft Model G-36 Serial Number E-3730
 Reg # N324P Owner's Name The Big Blessing LLC.

ITEM	DATE/TIME		INSPECTED BY	COMPLYING AGENCY	NOTATIONS	NEXT DUE
	DUE	CHECKED				
WOR 1 & 2 (Mick)	12-9	12-9-10	WAVNE C. MASSITT		+/- 0	



Duncan Aviation Inc.
 532 Beechcraft Ave.
 Chesterfield, MO 63005
 CRS# JQ03194F
 Phone (636) 536-7090 Fax (636) 536-7092

Hobbs: 815.9 REG #: N324P TYPE: Hawker Beechcraft G36 S/N: E-3730

- 1.) Tested and Inspected the static pressure systems, altimeter and automatic pressure altitude reporting system in accordance with 14 CFR Part 91.411 and ATC transponder tests and inspections in accordance with 14 CFR Part 91.413 and found to comply with 14 CFR part 43 appendix E and F. ATC Transponder P/N 011-00779-10 S/N 89102695, ADC P/N 011-00882-00 S/N 47807577, MFD/ Altimeter P/N 011-00972-03 S/N 88901843, Standby Altimeter P/N 58380069-3 S/N 459475 were tested to 20,000 feet on 6/17/2011.

The above described repair or maintenance is certified airworthy and approved for return to service. A detailed record is on file under Work Order # 0RX4A.

Signed [Signature] **DUNCAN**
 QI 1119
 For Duncan Aviation CRS# JQ03194F Date: 06/17/11



Duncan Aviation Inc.
 d/b/a Duncan Avionics Inc.
 532 Beechcraft Ave.
 Chesterfield, MO 63005
 CRS# JQ03194F
 Phone (636) 536-7090 Fax (636) 536-7092

Hobbs: 746.0 REG #: N324P TYPE: Beech G-36 S/N: E-3730

Reloaded Transponder software to G1000 system. Performed ATC transponder tests and inspections in accordance with 14 CFR Part 91.413 and found to comply with 14 CFR part 43 appendix E and F.

The above described repair or maintenance is certified airworthy and approved for return to service. A detailed record is on file under Work Order # 0L68A.

Signed [Signature] **DUNCAN**
 QI 1117
 For Duncan Aviation CRS# JQ03194F Date: 03/29/11

Altimeter Scale Correction Card

Standard Altitude Feet	Altimeter Reading
1,000	
0	-10
-500	490
1,000	990
1,500	1490
2,000	1980
3,000	2990
4,000	3990
5,000	4990
6,000	5990
8,000	7990
10,000	9990
12,000	11990
14,000	13990



**DUNCAN
AVIATION.**

Serial # 4780 1377
Part # 011-00380-02

Standard Altitude Feet	Altimeter Reading
16,000	15990
18,000	17990
20,000	19990
22,000	
24,000	
26,000	
28,000	
30,000	
32,000	
34,000	
36,000	
38,000	
40,000	
42,000	
44,000	
46,000	
48,000	
50,000	

Date 6/17/2011
Tested By [Signature]

INSTRUMENT, STATIC AND AVIONICS CHECKS RECORD OF COMPLIANCE

Aircraft Model _____ Serial Number _____

Reg # _____ Owner's Name _____

ITEM	DATE/TIME		INSPECTED BY	COMPLYING AGENCY	NOTATIONS	NEXT DUE
	DUE	CHECKED				

DUNCAN AVIATION

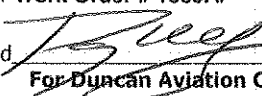



Duncan Aviation Inc.
18152 Edison Avenue
Hangar 2, Suite 250
Chesterfield, MO 63005
CRS# JQ03194F
Phone (636) 536-7090 Fax (636) 536-7092

Hobbs: 915.2 REG #: N324P TYPE: Hawker Beechcraft G36 S/N: E-3730

- 1.) Removed GIA #1 P/N 011-00781-01 S/N 48103881 and installed into #2 position. Removed #2 GIA P/N 011-00781-01 S/N 48103911 and installed into #1 position. Work performed and tested IAW Garmin G1000 System Maintenance Manual P/N 130-00422-03. No current U.S. Airworthiness Directives apply.

The above described repair or maintenance is certified airworthy and approved for return to service. A detailed record is on file under Work Order # 106JA.

Signed  **DUNCAN QI 1133**
For Duncan Aviation CRS# JQ03194F Date: 10/07/2011

<p>Aircraft Maintenance Record</p> <p>Tail Number: N324P Serial Number: E-3730 Work Order Number: 2784 Date: 5/9/13</p>		
<p><u>Perform 91.411/91.413 Checks:</u></p>		
<p>Performed tests required by 14CFR Part 91.413 in accordance with paragraph (c) Appendix E and Appendix F of 14CFR Part 43.</p>		
<p>#1 Transponder (GTX 33) - P/N- 011-00779-10 S/N- 89102695</p>		
<p>Altimeter(s), Altitude Reporting equipment and Static system required by 14CFR Part 91.411 have been performed in accordance with 14CFR Part 43 Appendix E.</p>		
<p>ADC (GDC-74) - P/N- 011-00882-00 S/N- 47807577 Standby Altimeter - P/N- 5934-3 S/N- 459475</p>		
<p>The aircraft and/or component listed on this WO 2784 were repaired and inspected I/A/W current FAA Regulations and are approved for return to service.</p>		
<p>Temple Electronics Co. Inc. CRS XDID608K SIGNED:  REPAIRMAN CERTIFICATE # 3129774</p>		